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Delayed-time domain impedance boundary conditions (D-TDIBC)

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Defining acoustically well-posed boundary conditions is one of the major numerical and theoretical challenges in compressible Navier–Stokes simulations. We present the novel Delayed-Time Domain Impedance Boundary Condition (D-TDIBC) technique developed to impose a time delay to acoustic wave reflection. Unlike previous similar TDIBC derivations (Fung and Ju, 2001–2004 [1,2], Scalo et al., 2015 [3] and Lin et al., 2016 [4]), D-TDIBC relies on the modeling of the reflection coefficient. An iterative fit is used to determine the model constants along with a low-pass filtering strategy to limit the model to the frequency range of interest. D-TDIBC can be used to truncate portions of the domain by introducing a time delay in the acoustic response of the boundary accounting for the travel time of inviscid planar acoustic waves in the truncated sections: it gives the opportunity to save computational resources and to study several geometries without the need to regenerate computational grids. The D-TDIBC method is applied here to time-delayed fully reflective conditions. D-TDIBC simulations of inviscid planar acoustic-wave propagating in truncated ducts demonstrate that the time delay is correctly reproduced, preserving wave amplitude and phase. A 2D thermoacoustically unstable combustion setup is used as a final test case: Direct Numerical Simulation (DNS) of an unstable laminar flame is performed using a reduced domain along with D-TDIBC to model the truncated portion. Results are in excellent agreement with the same calculation performed over the full domain. The unstable modes frequencies, amplitudes and shapes are accurately predicted. The results demonstrate that D-TDIBC offers a flexible and cost-effective approach for numerical investigations of problems in aeroacoustics and thermoacoustics.

1. Background

Direct Numerical Simulation (DNS) or Large Eddy Simulation (LES) have become standard approaches for high-fidelity simulations of unsteady compressible flows. In many cases, accurate flow predictions can be obtained only if the reflection of the acoustic waves at the boundaries is precisely defined and/or controlled [5–8]. Impedance is a widely-used quantity to characterize the reflection of acoustic waves at boundaries; it is naturally defined in the frequency domain whereas DNS/LES are performed in the time domain. Numerical methods permitting the imposition of impedance boundary conditions (IBCs)

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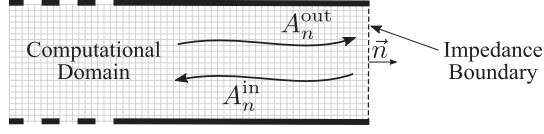


Fig. 1. Orientation of ingoing and outgoing acoustic waves A_n^{in} and A_n^{out} with respect to an Impedance Boundary (IB) with outward pointing normal \vec{n} .

in time domain solvers fall under the category of Time Domain Impedance Boundary Conditions (TDIBC). While existing TDIBC methods [1–4,9–14] have demonstrated their ability to achieve such goal, one case has not yet be considered in the Computational Aeroacoustics (CAA) community: a pure time delay. Developing Delayed-Time Domain Impedance Boundary Conditions (D-TDIBC), i.e. a TDIBC allowing to impose a time delay, is the goal of the present work.

The proper implementation of boundary conditions is critical for the stability of the numerical solution. For example, directly imposing mean values of velocity or pressure at the boundary leads to complete reflection of acoustic waves, hence leading to an artificial increase in disturbance energy.

Acoustic waves are the manifestation of perturbations around a mean state: the pressure field, for instance, can be expressed as $p(t) = p_0 + p'(t)$, where p_0 is the mean pressure and $p'(t)$ is the acoustic perturbation. In the present work, the time harmonic behavior $e^{+i\omega t}$ is assumed, yielding the convention $\phi(t) = \Re(\phi(\omega)e^{+i\omega t})$ for a generic periodic function $\phi'(t)$ oscillating with angular frequency ω . The corresponding Fourier transform convention reads:

$$\phi(\omega) = \int_{-\infty}^{+\infty} \phi(t) e^{-i\omega t} dt \quad (1)$$

with \mathcal{F} being the Fourier operator $\phi(\omega) = \mathcal{F}(\phi'(t))$, yielding the complex fluctuation amplitude $\phi(\omega)$.

The specific impedance $Z_*(\omega)$ is defined in the frequency domain as the non-dimensional ratio of complex pressure and the velocity component normal to the boundary [15]:

$$Z_*(\omega) = \frac{1}{\rho_0 c_0} \frac{p(\omega)}{\vec{u}(\omega) \cdot \vec{n}} \quad (2)$$

where ρ_0 and c_0 are the mean density and speed of sound, respectively. $Z_*(\omega)$ is a frequency-dependent complex-valued function, which is commonly used to characterize the response of an acoustic element. It is unbounded, contrary to the reflection coefficient R and the wall softness S , which are defined as:

$$R(\omega) = \frac{1 - Z_*(\omega)}{1 + Z_*(\omega)}, \quad S(\omega) = R(\omega) + 1. \quad (3)$$

Unlike the specific impedance Z_* , both the reflection coefficient R and the wall softness S are bounded in amplitude (for passive acoustic elements), making them the working variable of choice in computations. Fig. 1 illustrates the definition of the outgoing and ingoing acoustic waves, A_n^{out} and A_n^{in} , oriented in the same direction of and opposite to the outward normal \vec{n} . Following this convention, the outgoing and ingoing characteristic waves are defined as:

$$A_n^{\text{out}}(\omega) = \vec{u}(\omega) \cdot \vec{n} + \frac{p(\omega)}{\rho_0 c_0}, \quad A_n^{\text{in}}(\omega) = \vec{u}(\omega) \cdot \vec{n} - \frac{p(\omega)}{\rho_0 c_0} \quad (4)$$

Combining Eqs. (2) and (3) yields:

$$R(\omega) = \frac{A_n^{\text{in}}(\omega)}{A_n^{\text{out}}(\omega)} \quad (5)$$

$$S(\omega) = \frac{A_n^{\text{in}}(\omega) + A_n^{\text{out}}(\omega)}{A_n^{\text{out}}(\omega)} \quad (6)$$

Impedance, reflection coefficient and wall softness are defined in the frequency domain whereas DNS/LES are performed in the time domain. However, in order for the TDIBC to be physically admissible, the following three mathematical conditions must be fulfilled, as stated by Rienstra et al. [11]:

1. **Reality:** The characteristic waves imposed in the time domain should be purely real. Such a property can be ensured in the frequency domain, for instance in the reflection coefficient, if:

$$R^*(-\omega) = R(\omega) \quad (7)$$

where the superscript $*$ denotes the complex conjugate.

2. **Causality:** Any process should have a zero-response before it is triggered. Mathematically, such a condition is fulfilled if¹:

$$\Re(p_i) \leq 0 \text{ for } i \in [1; n_{\text{poles}}] \quad (8)$$

where p_i represents the i th complex pole of the reflection coefficient $R(\omega)$ prolonged to the Laplace plane and $\Re(\cdot)$ denotes the real part operator.

3. **Passivity:** The reflected wave amplitude should be lower or equal than the amplitude of the incident wave. This is fulfilled for:

$$|R(\omega)| \leq 1 \quad (9)$$

Should one of these constraints be violated, the imposition of TDIBC is expected to contribute to numerical instabilities. It is hence imperative to bear in mind such mathematical constraints when modeling $R(\omega)$ (or $S(\omega)$).

The time domain equivalent of $R(\omega)$ is obtained by use of the inverse Fourier transform $R(t) = \mathcal{F}^{-1}(R(\omega))$ and $S(t) = \mathcal{F}^{-1}(S(\omega))$. When dealing with the imposition of acoustic properties at a boundary condition, both the outgoing wave $A_n^{\text{out}}(t)$ (evaluated from the computational domain) and the reflection coefficient $R(\omega)$ (respectively the softness coefficient) are known. However, the ingoing wave $A_n^{\text{in}}(t)$ is unknown and has to be evaluated from the values of $A_n^{\text{out}}(t)$ and $R(\omega)$ (respectively $S(\omega)$). This can be done by recasting Eq. (5) (respectively Eq. (6)) and taking its inverse Fourier transform. As the inverse Fourier transform of a product yields a convolution integral [16,17], Eqs. (5) and (6) can be expressed in the time domain as follows:

$$A_n^{\text{in}}(t) = \int_0^t R(\tau) A_n^{\text{out}}(t - \tau) d\tau \quad (10)$$

$$A_n^{\text{in}}(t) = -A_n^{\text{out}}(t) + \int_0^t S(\tau) A_n^{\text{out}}(t - \tau) d\tau \quad (11)$$

where the bounds of the integrals in Eqs. (10) and (11) are reduced to $\tau \in [0, t]$ as we assume that $S(t)$, $R(t)$, $A_n^{\text{in}}(t)$ and $A_n^{\text{out}}(t)$ are defined on the interval $t \in [0, +\infty]$. For the causality constraint not to be violated, $A_n^{\text{in}}(t)$ must depend only on past and present values of $R(t)$ (respectively $S(t)$) and $A_n^{\text{out}}(t)$ but cannot depend on *future* values.

These assumptions restrict the bounds of the convolution integrals as it is the case when using the causal inverse Laplace transform [18,19]. Non coincidentally, the conversion from the frequency domain to time domain as done in most TDIBC technique (see section 3) requires the formal prolongation of the support of the Fourier transform, restricted to a real-valued frequency ω , to the complex domain. This step is hidden by an abuse of notation where the quantity $i\omega$ is intended as complex and hence following the formalism of Laplace transforms ($s = i\omega$).

Although mathematically equivalent to Eqs. (5) and (6), Eqs. (10) and (11) are not directly applicable in high-fidelity compressible solvers because the direct evaluation of the convolution integral requires a large amount of memory. Indeed, $A_n^{\text{out}}(t)$ and $R(t)$ need to be stored at every iteration and each boundary node or face. For simple geometries, Nottin [20,21] has solved Eq. (10) directly. In computationally intensive frameworks such as DNS/LES of turbulent flows, a more efficient compact-in-time method is required for the evaluation of $A_n^{\text{in}}(t)$, as shown in recent papers on this topic [1–4,14,22–26].

Advances in time domain impedance imposition in the electromagnetic community have driven novel acoustic boundary conditions for computational aeroacoustics [9,10,27,28]. Özyoruk et al. [10,27,29] and Tam and Auriault [9] have first developed formulations of TDIBC based on the z -transform and a Laurent series development of the impedance in the frequency domain resulting in a time domain Ordinary Differential Equation (ODE) problem. Numerical instabilities were observed by Özyoruk et al. [10,27,29] and discussed by Tam and Auriault [9]. Recent work on this topic suggests that these instabilities are due to the illposedness of the impedance boundary condition when coupled with Myers' slip flow velocity condition [11,30].

Fung and Ju [1,2] have proposed a causal formulation of TDIBC based on the wall softness S . As all Recursive Convolution methods (RC), Fung and Ju's method evaluates the convolution integral (Eq. (11)) in a low-storage manner. To avoid recomputing the whole convolution integral at all times (which would be increasingly costly in time) it is evaluated by (1) computing the change of the convolution integral's value over a time step and (2) store the updated value. Such method bypasses the need to store the time history of $S(t)$ and $A_n^{\text{out}}(t)$ as only the value of the convolution integral calculated up to the previous time step needs to be stored. The method relies on partial fraction modeling in the frequency domain, as discussed in Sec. 3.1. This modeling relies in identifying the poles and residues of a complex function – the wall softness coefficient S in Fung and Ju's case. The update of the convolution integral is done by solving Eq. (11) over a time step so that a low-cost evaluation is possible using a trapezoidal quadrature rule. Fung and Ju's formulation has been applied

¹ This is true when the $e^{+i\omega t}$ convention is adopted.

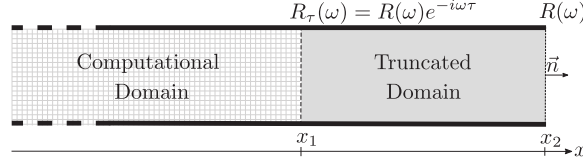


Fig. 2. Typical case where the computational domain can be cut at $x = x_1$. The propagation of acoustic waves in the truncated portion (between x_1 and x_2) is accounted for by the delayed reflection coefficient R_τ (Eq. (12)).

to one-dimensional inviscid Euler calculations [1,2] and, more recently, applied in fully compressible Navier–Stokes simulations [3,4], with some modifications required to ensure numerical stability of the implementation, namely setting the phase parameter to zero (see section 6.3 in Lin et al. [4]).

Another low-storage alternative to RC methods, is referred to in the literature as the Auxiliary Differential Equation (ADE) method. Such method also relies on partial fraction modeling but it involves the temporal integration of a set of first-order ordinary differential equations derived from the poles and residues equivalent to the convolution integrals in Eqs. (10) and (11). The major advantage of this method is that the same time-marching accuracy of the Navier–Stokes solver can be used for the TDIBC, thus ensuring consistency in the overall time accuracy order. A comparison between the ADE and the RC methods can be found in Dragna et al. [31]. The ADE method was also used to implement a TDIBC in Troian et al. [32].

In the combustion community, a state-space approach, called Characteristic Based state-space Boundary Condition (CBSBC), has been developed by Schuermans et al. [33], Kaess et al. [22] and Jaensch et al. [24], and used in Navier–Stokes simulations [14,22,24,26]. CBSBC relies on control theory of linear time-invariant (LTI) systems to build a reflection model through first-order differential matrix equations [24]. Two formulations of CBSBC were proposed. The first formulation of CBSBC is a state-space approach under the so-called controllable canonical form, i.e. based on the modeling of a transfer function as a rational polynomial. The same modeling procedure was used by Zhong et al. [13] to model the impedance. The transfer function represents the broadband reflection coefficient. Similarly to Fung and Ju’s method [1,2], this formulation of CSCBC aims at imposing wave reflection with near-zero time delay as, for practical reasons, it fails to handle pure time delay. Indeed, for the pure time delay imposition problem, CSCBC modeling requires a high number of Padé polynomial coefficients leading to ill-conditioned matrices, making this formulation difficult to use. A second formulation of CBSBC was proposed by Jaensch et al. [14] specifically to impose pure time delays. This approach consists in implementing a Linearized Euler Equation (LEE) solver for each of the impedance boundary conditions in the domain and performing the temporal integration using a first-order upwind scheme. The 1D spatial discretization needed by the LEE solver is done by storing in memory a state matrix of dimension $2N \times 2N$, and three matrices of dimension $2 \times N$, where N is the number of nodes considered in the spatial discretization. In their work, Jaensch et al. [14] have shown that to impose a pure time delay for a 1D wave propagation problem it was necessary to consider a number of nodes as high as $N = 1000$ in order to avoid acoustic energy dissipation.

In this work, a reformulation of Fung and Ju’s method based on the reflection coefficient is presented and used to impose pure time delays in DNS/LES. Although mathematically equivalent to the TDIBC multi-oscillator strategy used by Lin et al. [4], which targets the wall softness, the formulation adopted in the present work focuses on the reflection coefficient as it is a more natural and convenient choice for problems where delayed temporal responses of acoustic boundaries are of interest.

Section 2 will present the definition of delayed reflection, which allows truncating long portions of the computational domains. Section 3 will introduce the TDIBC formulation used in this work and the fitting technique necessary to model a delayed reflection coefficient. In Sec. 4 D-TDIBC will be used to impose a time delay in a 1D wave propagation test case. Finally, in Sec. 5 a 2D thermoacoustically unstable combustion chamber is studied via DNS, using the D-TDIBC technique to reduce the domain size.

2. Problem formulation: delayed impedance boundary conditions

In many simulations of industrial systems, longitudinal waves propagate over elongated portions of the computational domain (e.g. in the exhaust pipe of a car engine, in the chimney on an industrial furnace or, in general, for ducts of both constant section and speed of sound), merely resulting in a time lag of the acoustic response. In such cases the computational domain can be significantly reduced, by replacing such elongated portions with a boundary condition accounting for the temporal delay associated with the acoustic wave propagation time.

Fig. 2 shows an acoustic problem in which the portion from x_1 to x_2 is truncated (grey area). In this computational setup, the boundary condition located in $x = x_1$ must account for the acoustic properties of the truncated domain. This is possible if the boundary condition at $x = x_1$ imposes a *time delay* to the wave reflection located at $x = x_2$: the resulting reflection coefficient is called *delayed reflection coefficient* [7,34]. Assuming planar and linear wave propagation yields the following expression for the delayed reflection coefficient R_τ :

$$R_\tau = R(\omega)e^{-i\omega\tau} \quad \text{with} \quad \tau = 2(x_2 - x_1)/c_0 \quad (12)$$

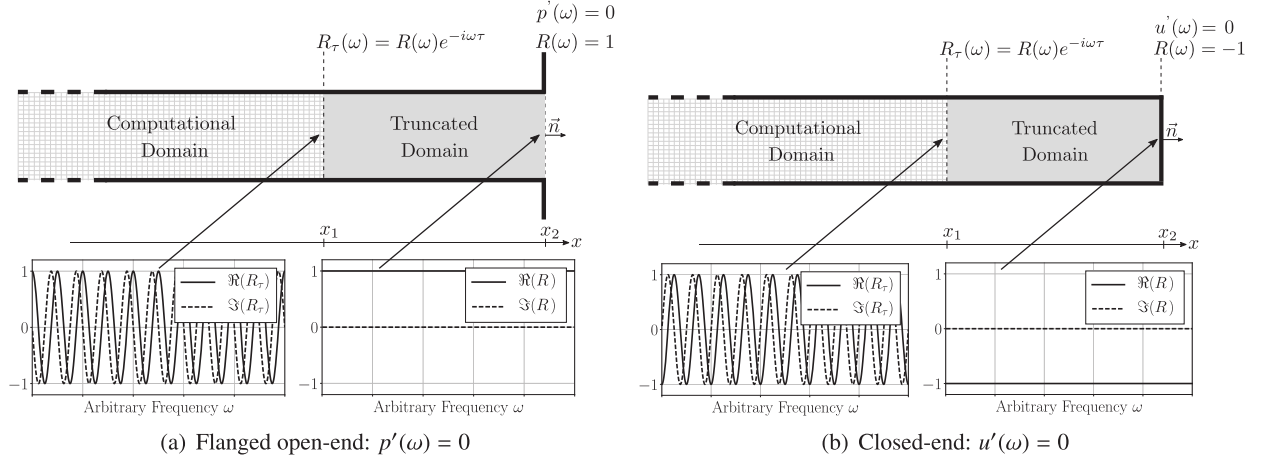


Fig. 3. Acoustic variables for two canonical limit cases: (a) the flanged open-end, and (b) the closed-end (hard wall) cases. Both configurations are fully reflective and have a purely real reflection coefficient R at all frequencies in $x = x_2$. The truncation of the grey areas can be modeled by the imposition of a time delay τ . The resulting delayed reflection coefficient R_τ at $x = x_1$ is complex. At the bottom of the figure both real (solid lines) and imaginary parts (dashed lines) of R_τ and R are shown for both cases.

where τ is the total time delay (i.e. the time required by a wave to travel from x_1 to x_2 and back) and $i = \sqrt{-1}$ is the imaginary unit. In this formulation of the delayed reflection coefficient, it is assumed that the base pressure in the truncated section is uniform. An different formulation should be derived for cases where, for example, significant pressure drops are present in the truncated section.

Fig. 3 illustrates the impact of an arbitrary time delay τ on two classical limit cases: the flanged open-end, i.e. $p(\omega) = 0$ (Fig. 3(a)), and the closed-end, i.e. $u(\omega) = 0$ (Fig. 3(b)). These cases have purely real reflection coefficients $R(\omega)$ at the acoustic boundary located in $x = x_2$. For all frequencies ω , the flanged open end corresponds to a reflection coefficient of $R(\omega) = 1$ and the closed end to $R(\omega) = -1$. The delayed reflection coefficient $R_\tau(\omega)$ becomes complex-valued and frequency-dependent even though $R(\omega)$ is purely real and frequency-independent.

In this work we build upon the multi-oscillator TDIBC methodology of Fung and Ju [2], with the phase-parameter correction by Lin et al. [4], to create a novel TDIBC formulation, called Delayed-Time Domain Impedance Boundary Condition (D-TDIBC), relying on the determination of the partial fraction representation (via poles and residues) of the delayed complex reflection coefficient, as outlined in Sec. 3.1. It is important to note that by adopting Fung and Ju's formalism, which focuses on the causal reconstruction of the softness (or absorption) coefficient, it is indeed appropriate to talk about a multi-oscillator fitting strategy, where the fitted softness is the linear superposition of the softnesses of various causal oscillators, which can be each representative of real acoustically absorptive panels; on the other hand, the decomposition of the reflection coefficient in partial fractions as done in the present work (see Eq. (13)) is not tied to a particular physical interpretation. Due to the peculiar mathematical nature of the delayed complex reflection coefficient (Eq. (12), plotted in Fig. 3), exhibiting periodic variations in the frequency domain, the proposed D-TDIBC strategy was developed to handle a large number of poles and residues. While the focus herein is on the reflection coefficient R , the same fitting methodology outlined in Sec. 3 can be extended to the wall softness S .

3. Methodology

3.1. Time domain imposition of complex reflection coefficient

The direct numerical integration of Eq. (10) does, in principle, allow to impose a complex reflection coefficient in the time domain; however, this results in rapidly unsustainable memory and CPU costs, especially in DNS/LES simulations. Building upon the methodologies of Fung and Ju [1,2], Scalo et al. [3] and Lin et al. [4], we approximate a generic target reflection coefficient $R(\omega)$ with the reflection coefficient of the boundary condition R_{BC} expressed as a sum of rational functions:

$$R(\omega) \simeq R_{BC}(\omega, n_0) = \sum_{k=1}^{2n_0} \frac{\mu_k}{i\omega - p_k} \quad (13)$$

where $R_{BC}(\omega, n_0)$ is the reflection coefficient model and the poles and residues (p_k, μ_k) must come as n_0 conjugate pairs:

$$p_{2k} = p_{2k-1}^*; \quad \mu_{2k} = \mu_{2k-1}^*; \quad k \in [1; n_0] \quad (14)$$

These conjugate pairs ensure that the reflection coefficient R is real-valued in the time domain. The time domain equivalent of Eq. (13) is:

$$R_{BC}(t, n_0) = \sum_{k=1}^{2n_0} \mu_k e^{p_k t} H(t) \quad (15)$$

where the Heaviside function $H(t)$ indicates that the causality constrained is fulfilled. The time-domain incoming wave value $A_n^{\text{in}}(t)$ corresponding to $R_{BC}(\omega, n_0)$ is obtained by substituting Eq. (15) into Eq. (10) yields

$$A_n^{\text{in}}(t) = \sum_{k=1}^{2n_0} \int_0^t \mu_k e^{p_k \tau} A_n^{\text{out}}(t - \tau) d\tau = \sum_{k=1}^{2n_0} I_k(t) \quad (16)$$

comprising $2n_0$ convolution integrals called I_k , which can be split into two contributions [1,2]:

$$I_k(t) = I_k(t - \Delta t) e^{p_k \Delta t} + \mu_k \int_{t-\Delta t}^t e^{p_k \tau} A_n^{\text{out}}(t - \tau) d\tau \quad (17)$$

Equation (17) shows that each $I_k(t)$ can be evaluated recursively: the first term corresponds to the temporal integration over the interval $\tau \in [0, t - \Delta t]$ and the second term to the temporal integration over the interval $\tau \in [t - \Delta t, t]$. It provides a low memory storage method to compute the ingoing wave $A_n^{\text{in}}(t)$. The only quantities needed to be stored are $I_k(t - \Delta t)$ and $A_n^{\text{out}}(t)$. The resulting CPU and memory cost does not increase with the number of time steps, and the computational overhead associated with the integral in Eq. (17) over a time step Δt is minimal.

We adopt a trapezoidal quadrature rule for the discretization of the integral in Eq. (17), yielding [1]:

$$I_k(t) = I_k(t - \Delta t) e^{p_k \Delta t} + \alpha_k A_n^{\text{out}}(t) + \beta_k A_n^{\text{out}}(t - \Delta t) \quad (18)$$

$$\alpha_k = \mu_k \left(\frac{e^{p_k \Delta t} - 1}{p_k^2 \Delta t} - \frac{1}{p_k} \right); \quad \beta_k = \mu_k \left(\frac{e^{p_k \Delta t} - 1}{p_k^2 \Delta t} - \frac{e^{p_k \Delta t}}{p_k} \right) \quad (19)$$

It should be noted that Eqs. (18) and (19) limit the overall temporal accuracy order to second order; this can be overcome by developing an alternative time integration strategy based on the Auxiliary Differential Equation (ADE) method (see Background), which is out of the scope of the present paper.

The initial values of $I_k(t)$ and A_n^{out} are set to zero in the simulations [1–4]. However, one can make a checkpoint restart by initializing the $I_k(t)$ using stored values from a previous simulation. Using Eqs. (18) and (19), complex impedances can be imposed in DNS/LES solvers.

The methodology to determine the model inputs (μ_k, p_k) is now presented.

3.2. Multi-Pole Modeling

As discussed in Sec. 3.1, in order to impose the reflection coefficient $R(\omega)$ we need to determine the set of poles and residues (p_k, μ_k) . There is no evidence of the uniqueness of the solution for a set of (μ_k, p_k) . We propose here a method to determine the poles and residues that is viable when modeling delayed reflection coefficients (Eq. (12)). It is a tedious task as the delayed reflection coefficient $R_\tau(\omega)$ has many zeros introduced by the complex exponential $e^{-i\omega\tau}$. The method consists in an iterative least-square fit. At each iteration, a rational fraction is added to the model until the model reflection coefficient $R_{BC}(\omega)$ converges to the desired reflection coefficient $R(\omega)$.

In order to develop this fitting method, we will first describe the properties of the rational fractions in Eq. (13).

3.2.1. Pole Base Function properties

The reflection coefficient of the boundary condition $R_{BC}(\omega, n_0)$ is a sum of n_0 rational fractions, called the Pole Base Functions (PBF), denoted R_k :

$$R_{BC}(\omega, n_0) = \sum_{k=1}^{n_0} R_k(\omega) = \sum_{k=1}^{n_0} \underbrace{\left[\frac{\mu_k}{i\omega - p_k} + \frac{\mu_k^*}{i\omega - p_k^*} \right]}_{\text{Pole Base Function}} \quad (20)$$

Equation (20) can be recast by expressing the poles and residues in their algebraic forms, i.e. as $\mu_k = a_k + ib_k$ and $p_k = c_k + id_k$ with $a_k, b_k, c_k, d_k \in \mathbb{R}$. Each R_k is causal if $c_k < 0$ [11,35].

Although Fung & Ju [2] suggested that the phase parameter $B_k = b_k d_k + a_k c_k$ could be not null, Lin et al. [4] have experienced spurious oscillations, leading to unstable numerical simulations, when the parameter B_k was unconstrained $B_k \neq 0$. Although the TDIBC of interest is based on the reflection coefficient R rather than on the wall softness coefficient S , similar behavior was observed in preliminary numerical trials by the authors. Thus, the phase parameter was constrained to $B_k = 0$, i.e. $b_k = -a_k c_k / d_k$, leading to:

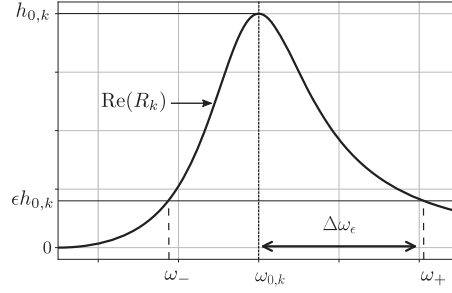


Fig. 4. Real part of a typical PBF R_k (Eq. (21)). Three properties are visualized: $\omega_{0,k}$, $h_{0,k}$ and $\Delta\omega_\epsilon$. At the resonant angular frequency $\omega_{0,k}$, the real part of R_k is maximum. The peak height $h_{0,k}$ is the value of $R_k(\omega)$ at $\omega = \omega_{0,k}$. For a given percentage ϵ ($\epsilon \in [0, 1]$), the half-width $\Delta\omega_\epsilon$ is defined on the right-hand-side. ω_- and ω_+ are the solutions of the equation $\Re(R_k) = \epsilon$ and are used to define the width.

$$R_k(\omega) = \frac{2a_k i \omega}{-\omega^2 - 2c_k i \omega + (c_k^2 + d_k^2)} \quad (21)$$

There are three degrees of freedom in Eq. (21), namely a_k , c_k and d_k . The objective of this section is to link these degrees of freedom to the properties of the PBF.

A typical PBF is presented in Fig. 4 where three properties of the PBF can be identified: $\omega_{0,k}$, $h_{0,k}$ and $\Delta\omega_\epsilon$. $\omega_{0,k}$ is the frequency where the real part of R_k is maximum: it is referred as the *resonant frequency*. $h_{0,k}$ is the peak height, that is, the value of $R_k(\omega)$ at $\omega = \omega_{0,k}$. The last property shown in Fig. 4 is the half-width $\Delta\omega_\epsilon$ defined for a given percentage ϵ of $h_{0,k}$ ($\epsilon \in [0, 1]$). Because the PBF is not symmetric, at a given percentage ϵ , two half-widths $\Delta\omega_\epsilon$ can be defined: one on the left-hand side (using ω_-) and one on the right-hand side (using ω_+). A choice is made to consider only the right half-width shown in Fig. 4, hence overestimating the width.

One can show that [3,4]:

$$\omega_{0,k}^2 = c_k^2 + d_k^2 \quad (22)$$

Equation (22) gives a first constraint, say on d_k . Injecting Eq. (22) into Eq. (21) and evaluating the resulting equation at $\omega = \omega_{0,k}$ yields:

$$R_k(\omega_{0,k}) = h_{0,k} = -\frac{a_k}{c_k} \quad (23)$$

Using the constraints in Eqs. (22) and (23), R_k becomes:

$$R_k(\omega) = \frac{2h_{0,k}c_k i \omega}{\omega^2 + 2c_k i \omega - \omega_{0,k}^2} \quad (24)$$

In Eq. (24), c_k is the only remaining degree of freedom. Fig. 5(a) illustrates its effect on the width of a PBF R_k for a peak height $h_{0,k} = 1$ at the resonant frequency $f_{0,k} = \omega_{0,k}/2\pi = 100$ Hz. In the case of a pure delay the width is known: $R(\omega)$ is a periodic function of period $T = 2\pi/\tau$ (Fig. 5(b)). As shown in Fig. 5(a), the real part of R_k goes to zeros for $\omega \ll \omega_{0,k}$ and $\omega \gg \omega_{0,k}$. A consequence of this property is that a single PBF can fit a single peak between two consecutive zeros. For the pure delay the frequency range between two consecutive zeros is $T/2$. To adjust the width to the pure delay case, we need to find the value of c_k for a given ϵ such that:

$$\Re \left[R_k \left(\omega_{0,k} + \frac{T}{4} \right) \right] = \Re \left[R_k \left(\omega_{0,k} + \frac{\pi}{2\tau} \right) \right] = \epsilon \quad (25)$$

where $\Re(\cdot)$ and $\Im(\cdot)$ are the real and imaginary parts, respectively. Solving for c_k in Eq. (25), we obtain the criterion for a given time delay τ :

$$c_k(\epsilon, \omega_{0,k}) = -\frac{\tau \omega_{0,k} + \frac{\pi}{4}}{2\pi \omega_{0,k} + \tau} \sqrt{\frac{\epsilon}{1-\epsilon}} \quad (26)$$

Equations (22), (23) and (26) give constraints allowing to control the height and the width of a PBF for any resonant angular frequency $\omega_{0,k}$. These constraints are used in the algorithm allowing the determination of the set of poles and residues (p_k, μ_k) presented in the next section.

3.3. Iterative Multi-Pole Modeling Technique

Now that the properties of a single PBF are known, it is possible to go back to Eq. (20) where a sum of n_0 PBFs are used to match a reflection coefficient $R(\omega)$. In the case of pure delay, it is a tedious task as the complex exponential function, introduced by the time delay (see Eq. (12)), has a large number of zeros.

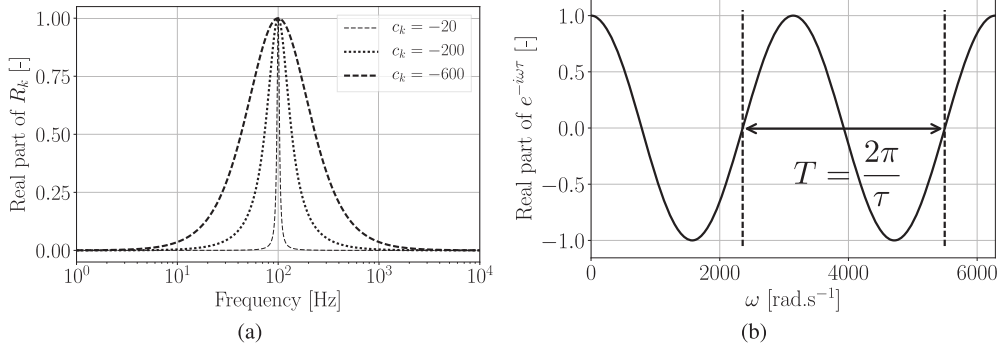


Fig. 5. (a) Influence of the c_k parameters for $f_{0,k} = \frac{1}{2\pi}\omega_{0,k} = 100$ Hz and $h_{0,k} = 1$. (b) Real part of a purely delayed reflection coefficient R_τ for a time delay τ . The real part is a periodic function (cosine) of period $T = \frac{2\pi}{\tau}$.

A common practice to obtain the multi-pole approximation of a complex function is to use the Vector Fitting method [36, 37]. It was used to model TDIBC using partial fractions by several authors [12,31,38–40]. However, preliminary tests have shown only limited results when applied to complex exponentials as convergence could not be reached when modeling pure time delays. To tackle this issue an iterative fitting technique has been devised.

The least-square fit algorithm is used to minimize the distance of the target function $R(\omega)$ to the fit function $R_{BC}(\omega)$ (Eq. (21)) on both real and imaginary parts. The global least-square residual ξ is defined as the sum of the squared distance between $R(\omega)$ and $R_{BC}(\omega)$:

$$\xi = \xi_R + \xi_I \quad (27)$$

where ξ_R and ξ_I are the least-square residuals on the real and imaginary parts, respectively. They are defined as the squared sum of the point-to-point oriented distance, E_R and E_I , between $R(\omega)$ and $R_{BC}(\omega)$:

$$\xi_R = \sum_{i=1}^m E_R^2(\omega_i) = \sum_{i=1}^m \Re \left[R(\omega_i) - R_{BC}(\omega_i) \right]^2 \quad (28)$$

$$\xi_I = \sum_{i=1}^m E_I^2(\omega_i) = \sum_{i=1}^m \Im \left[R(\omega_i) - R_{BC}(\omega_i) \right]^2 \quad (29)$$

where m is the number of discrete values taken into account in the frequency array, i is the index of the points, $R(\omega_i)$ is the value of the target reflection coefficient at the i th point of the discrete angular frequency array ω_i and $R_{BC}(\omega_i)$ is the value of the fit function at the i th point.

The fitting procedure is presented in the Algorithm 1. It starts with only one term: $n_0 = 1$ in Eq. (20) (line 1 in the Algorithm 1). The conditional loop iterates until the number of PBF n reaches the final value $n = n_0$. In line 3, we seek for the frequency where the error on the real part is maximum. The resonant frequency of R_n is then set to this frequency (line 4) and its peak height $h_{0,n}$ is chosen to cancel the error on the real part at that point (line 5). The c_n parameter is initialized using $\epsilon = 1\%$ in Eq. (26). Equations (22) and (23) are used to determine the values of d_n and a_n (line 7 and 8). Finally, the pole and residue (p_n, μ_n) of the n th PBF are initialized (lines 9 and 10). The optimization stage (line 11) minimizes the least-square residual ξ (see Eq. (27)) and all of the values of the parameters p_k and μ_k are optimized for $k \in [1; n]$. Finally, the order of the model n is incremented so that an additional PBF can be added (line 12).

An example of iterative fit is presented in Sec. 4.2 where the model $R_{BC}(\omega)$ is shown at several iterations of the Algorithm 1.

Neither the uniqueness of the solution for a set of (p_k, μ_k) nor the convergence of the Algorithm 1 have been mathematically proven. Such derivations are beyond the scope of this work and require further studies. In practice, the convergence is left at the mercy and robustness of the least square algorithm, which is very reliable, and convergence was always reached for the delayed reflection coefficients considered in this study.

The coefficients $1/c_k$ and $1/d_k$ are time constants, and, as such, they might constrain the maximum allowed time step size of the overall computation due to numerical stability if either $|c_k \Delta t|$ or $|d_k \Delta t|$ is too large [12]. In the proposed fitting method, there is no explicit constraint for such parameters. However, in practice these methods are used in explicit DNS and LES codes where Δt is very small so that the no numerical stability issues were encountered by the authors while using poles and residues provided by Algorithm 1.

Algorithm 1: Preprocessing for D-TDIBC: find (p_k, μ_k) in Eq. (20) to model the target function $R(\omega)$.

```

Input :  $n_0, \omega_i, R(\omega_i), \tau$ 
Output:  $(p_k, \mu_k)$  for  $k \in [1, n_0]$ 

1  $n \leftarrow 1$ 
2 while  $n < n_0$  do
3   Find  $\omega_{\max}$  such that the error on the real part is maximum  $\equiv \{\omega_{\max} \mid \forall \omega_i : E_R(\omega_i) \leq E_R(\omega_{\max})\}$ 
4   Set  $\omega_{0,n} = \omega_{\max}$  using Eq. (22)
5   Set the resonant peak height to cancel the error on the real part  $h_{0,n} = E_R(\omega_{\max})$  using Eq. (23)
6   Set the parameter  $c_n$  using Eq. (26)
7   Set the parameter  $d_n$  using Eq. (22)
8   Set the parameter  $a_n$  using Eq. (23)
9    $\mu_n \leftarrow a_n + i(-d_n c_n / d_n)$ 
10   $p_n \leftarrow c_n + i d_n$ 
11  Least-square fit: modify the values of  $p_k$  and  $\mu_k$  for  $k \in [1, n]$  to minimize  $\xi$  (see Eq. (27))
12   $n \leftarrow n + 1$ 
13 end

```

4. Validation for purely reflective time-delayed impedance

Section 3 has provided the modeling methodology necessary to account for acoustic delays. In this section, the objective is twofold: (1) to demonstrate the applicability of the modeling procedure on a limit case: the delayed pure reflection, (2) to validate the ability of TDIBC to impose a time delay to account for acoustic wave propagation in the truncated portion of the domain. In Sec. 4.1 we present the one-dimensional numerical setup used for validation. The methodology proposed in Sec. 3 will be used to model a reflection coefficient corresponding to a pure delay. In Sec. 4.3 a time domain simulation is used to demonstrate that D-TDIBC imposes the correct time delay τ , wave amplitude and phase.

4.1. Test case presentation

The aim of the simulation is to study the propagation of a Gaussian acoustic wave in the domain Ω defined on $x \in [-1, 1.75]$ m (Fig. 6(a)). The mean speed of sound is $c_0 = 350 \text{ m s}^{-1}$. This perturbation will propagate on the positive x direction. The physical domain is split into two sub-domains: $\Omega = \Omega_c + \Omega_m$. The computational domain Ω_c is $x \in [-1, 1]$ m and the modeled domain Ω_m is $x \in [1, 1.75]$ m (gray area in Fig. 6(a)). The D-TDIBC is used to model the wave propagation in Ω_m . At the left boundary condition ($x = x_l = -1$ m) the reflection coefficient is $R_l = 1$ and at the right boundary condition ($x = x_r = 1.75$ m) the reflection coefficient is $R_r = -1$. In the computational domain, the reflection coefficient at $x = 1$ m accounting for R_r has to be modeled by a time delay (see Sec. 1) leading to $R_r|_{x=1 \text{ m}} = -e^{-i\omega\tau}$. The time delay corresponding to this modeled domain is $\tau = 2(x_r - x_{BC})/c_0 = 4.28$ ms. The acoustic time for the initial wave located in x_0 to travel back at its initial position is $T = 2(x_r - x_0)/c_0 = 10.0$ ms.

4.2. Delayed impedance modeling

To impose a pure time delay, the D-TDIBC method first requires to build a model for the delayed reflection coefficient using the Algorithm 1 (see Sec. 3). In theory, the delayed reflection coefficient $R_r|_{x=1 \text{ m}} = -e^{-i\omega\tau}$ must be modeled up to infinite frequencies. In practice, it is not feasible as it would require an infinite number of PBFs. In numerous applications, the frequencies of interest are limited to a given range (Fig. 6(b)) and it is sufficient to model $R(\omega)$ below a cutoff frequency f_c . For $f > f_c$, a non-reflecting boundary condition is used. For example, for the setup presented in Fig. 6(a), the frequency content of the initial Gaussian wave is below 1 kHz so that the chosen cutoff frequency is $f_c = 1$ kHz. A low-pass filter is applied to the theoretical delayed reflection by multiplying the delayed reflection coefficient by an envelope function ψ allowing a smooth transition from 1 to 0 over a frequency range:

$$\psi(f) = \frac{1}{2} \left[1 - \tanh \left(\frac{f - f_c}{\delta} \right) \right] \quad (30)$$

where f is the frequency and δ is a constant that specifies the frequency range of the transition from 1 to 0 (see Fig. 6(b)). The envelop function ψ keeps the phase of the delayed reflection coefficient unchanged. The target function to be modeled is then:

$$R(\omega) = \psi(f) \times R_r|_{x=1 \text{ m}} \quad (31)$$

Fig. 7 shows the model at iterations 1, 2, 5 and 20 of the Iterative Multi-Pole Modeling Technique (Algorithm 1). $R(\omega)$ is the filtered delayed reflection coefficient shown in Fig. 6(b) and $R_{BC}(\omega)$ is the reflection coefficient model at a given iteration defined in Eq. (13). Their real and imaginary parts are shown in the top and bottom plots, respectively. At each iteration,

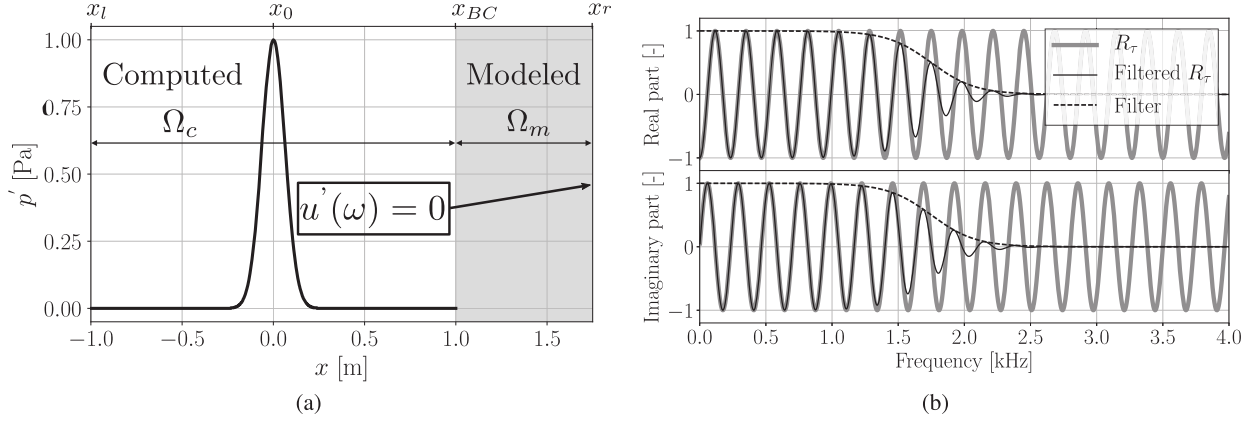


Fig. 6. (a) Initial solution and visualization of the computational (Ω_c) and modeled (Ω_m) domains. (b) Theoretical (gray solid line) and filtered (black solid line) delayed reflection coefficients. The filter shape (Eq. (30)) is shown (black dashed line).

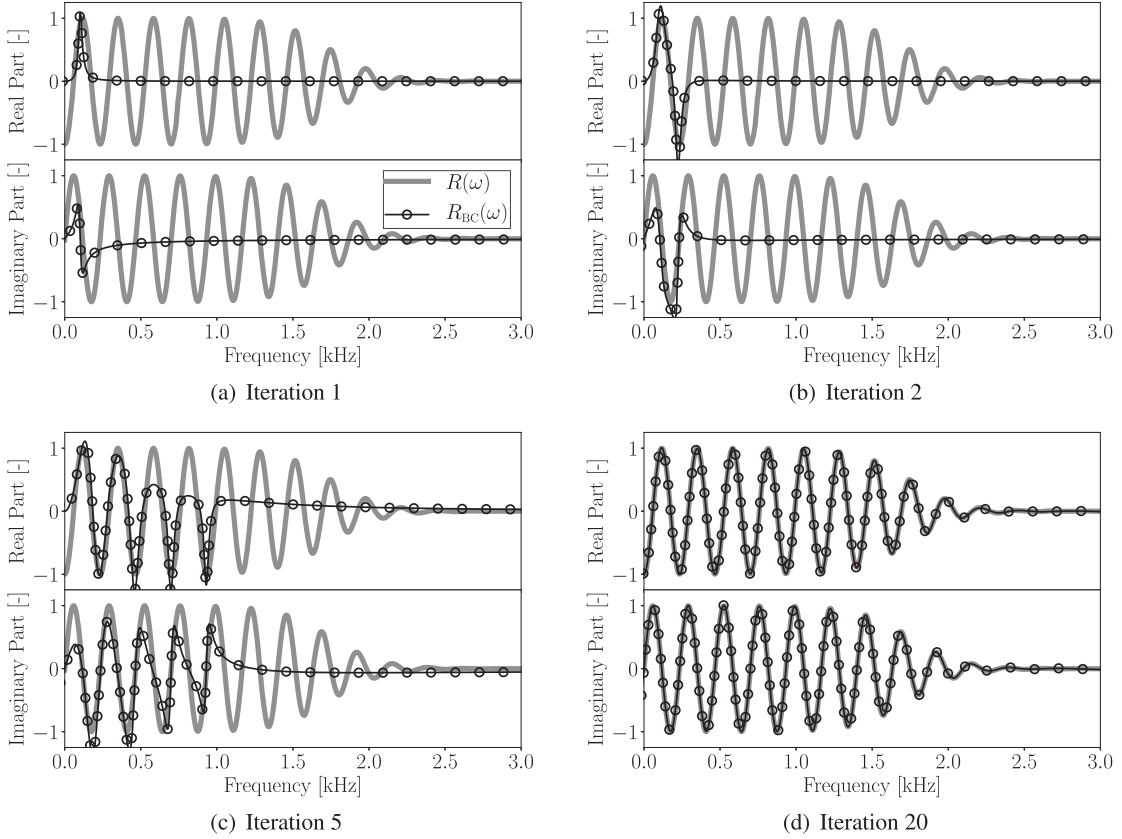


Fig. 7. Reflection coefficient $R_{BC}(\omega)$ modeled by the Iterative Multi-Pole Modeling Technique algorithm (Algorithm 1 in Sec. 3.3) a several iterations. An accurate model is obtained for $n_0 = 20$, where n_0 is the number of PBF R_k as in Eq. (20).

the method finds the biggest point-to-point oriented distance between the real parts (Eq. (28)) of $R(\omega)$ and $R_{BC}(\omega)$. Fig. 7(a) shows the model at the end of the first iteration. The initial solution of $R_{BC}(\omega)$ targets the first peak in the real part where $\Re(R(\omega)) = 1$. In Fig. 7(b), the targeted peak is where $\Re(R(\omega)) = -1$. At an iteration i , initial guesses for p_i and μ_i are found to target a single peak at the time. Fig. 7 illustrates how the model propagates from low to high frequencies. The algorithm is considered converged after 20 iterations as the maximum point-to-point error on the modulus is below 1%, where we define the error as $E = \max(|R(\omega) - R_{BC}(\omega, n_0 = 20)|)$. The D-TDIBC model used here is, thus, made of 20 poles and residues (p_k, μ_k), that is to say only 40 complex constants (cf. Appendix A).

At $\omega = 0$, the modulus of the reflection coefficient to be modeled is $|R(\omega = 0)| = 1$. Such a condition cannot be fulfilled by the D-TDIBC as each of the PBF has a null modulus at the zero-frequency limit: $|R_k(\omega = 0)| = 0$. However, the frequency

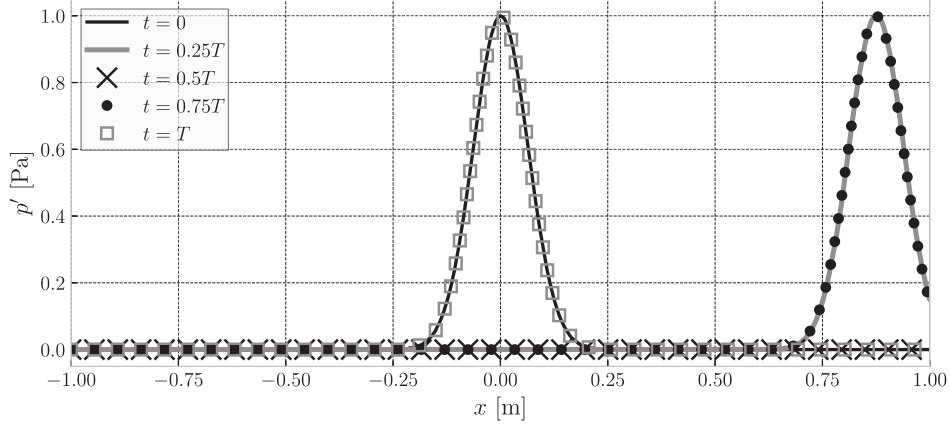


Fig. 8. Propagation and reflection of a Gaussian pressure wave in the domain $\Omega_c + \Omega_m$ by simulating only Ω_c . The acoustic properties of Ω_m are imposed using D-TDIBC.

at which the modulus of the reflection coefficient modeled by D-TDIBC becomes zero can be made arbitrarily low in the fitting stage. The modulus of the modeled reflection coefficient $R_{BC}(\omega)$, presented in Fig. 7(d), tends to zero for frequencies below 0.1 Hz.

The reflection coefficient modeled by the poles and residues fulfills the causality and reality conditions by construction of the method. However, it is interesting to note that for a small number of poles and residues (hence for a low accuracy of the fit) the passivity condition is violated, as shown in Fig. 7(a) where $|R(\omega)| > 1$ around 0.15 kHz. Nonetheless, for a higher number of poles and residues, the passivity condition is also verified.

4.3. One-dimensional wave propagation

In this section, the behavior of the D-TDIBC is investigated first on a basic one-dimensional wave propagation problem. Then the time domain response of each PBF is inspected to highlight the mechanism of D-TDIBC.

The one-dimensional wave propagation is computed using the AVBP solver. AVBP is a three-dimensional fully compressible Navier–Stokes equation solver. A two-step Taylor–Galerkin scheme, called TTGC is used. TTGC is third-order accurate in space and time [41,42]. Characteristics boundary conditions (NSCBC) are used [5,8,43,44]. In the NSCBC framework the characteristic waves are evaluated at the boundaries. TDIBC prescribes the ingoing characteristic wave from the outgoing characteristic wave and is, thus, consistent with the NSCBC formalism.

As the reflection coefficient modeled in x_r is $R_r = -1$, the wave is *a priori* expected to be fully reflected: the amplitude of the reflected wave must be equal to the one of the incident wave. As R_r is real-valued, the reflected wave is expected to be centered in $x = 0$ m after a time $t = T$ (see Sec. 4.1).

Fig. 8 illustrates the propagation of the pressure wave from time $t = 0$ up to $t = T$. At $t = 0$ the pressure field corresponds to the initial solution shown in Fig. 6(a). At $t = 0.25T$ the wave has propagated in the positive x direction and is crossing the boundary condition at $x = x_{BC} = 1$ m. At $t = 0.5T$ the pressure wave has completely left the computational domain. This result is expected: at time $t = 0.5T$ the wave is propagating inside the modeled domain Ω_m , i.e. the truncated portion (gray area in Fig. 6(a)). If one were to compute the complete domain, at $t = 0.5T$ the acoustic pressure $p'(x)$ would be zero for $x \in [-1, 1]$ m. At $t = 0.75T$ the pressure wave is re-injected in the computational domain by the boundary condition. The reflected pressure wave has the same amplitude than the incident wave (1 Pa) as expected. At $t = T$ the reflected Gaussian wave is centered at $x = 0$ m which means that the time delay is precisely prescribed by D-TDIBC.

The acoustic pressure and velocity at the boundary are recorded at the boundary condition at $x_{BC} = 1$ m during the simulation. The results are shown in Fig. 9(a). Additionally, the acoustic energy E_a contained in the domain is shown. For the sake of clarity, each variable is normalized by its maximum. At first no acoustic activity is seen as the wave propagates inside the domain, the acoustic energy is maximum.

The acoustic pressure p' at the boundary in Fig. 9(a) is consistent with the results in Fig. 8. At first no acoustic pressure is seen by the boundary until the Gaussian wave crosses the boundary at $t \simeq 0.25T \simeq 2.5$ ms. After the wave has crossed the boundary, the acoustic pressure returns to zero. After the time delay τ (Fig. 9(a)) at $t \simeq 0.75T \simeq 7.5$ ms the reflected wave is re-entering the computational domain with the same amplitude as the incident wave. Finally, the acoustic pressure p' returns to zero. The amplitude of the reflected acoustic velocity u' ($t \simeq 0.75T \simeq 7.5$ ms) is also the same as the incident wave ($t \simeq 0.25T \simeq 2.5$ ms) although the sign is changed: u' is negative. This indicates that the wave is traveling in the negative x direction after the reflection at $x_r = 1.75$ m. This result is consistent with the imposition of a reflection coefficient $R_r = -1$ in x_r . When the wave crosses the boundary the acoustic energy initially contained in the domain is lost due to the acoustic flux at the boundary condition in x_{BC} . The acoustic energy contained in the domain retrieves its initial value when

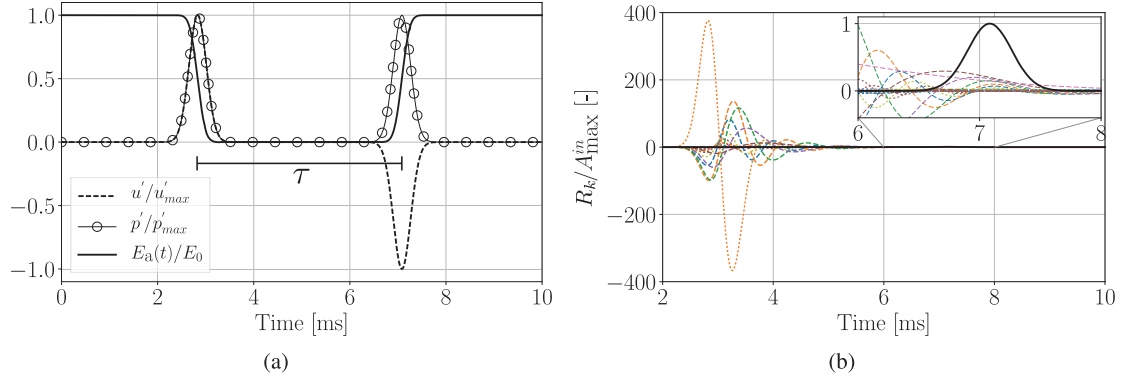


Fig. 9. (a) Acoustic velocity (dashed line) and pressure (solid line with circles) at the boundary condition. (b) Temporal response of 20 PBF (dashed and dotted lines) and of the ingoing wave imposed by the D-TDIBC (black solid line).

the reflected wave re-enters the domain. This stresses the fact that D-TDIBC conserves the acoustic energy while imposing a time delay.

In their work, Jaensch et al. [14] solve a similar problem: they impose a pure time delay at a boundary condition with a one-dimensional incident Gaussian wave. As discussed in Sec. 1, their method uses a first order upwind numerical scheme to discretize the one-dimensional LEE over the truncated domain. Their method is, thus, inherently dissipative and the dissipation increases linearly with the cell size of the spatial discretization. In order to achieve a dissipation level comparable to the one obtained here, it was necessary to use 1000 points. As a consequence, in the state-space model used, the state matrix is of dimension 2000×2000 , i.e. 4 million scalar values. D-TDIBC seems to yield comparable results for a lower memory storage requirement as only 40 complex constants were used to model the delayed reflection coefficient.

Fig. 9(b) illustrates the contribution of each PBF R_k . The ingoing wave $A_n^{\text{in}}(t)$ imposed by D-TDIBC is the sum of all the R_k . Before $t \approx 7$ ms the ingoing wave is null: the PBFs are canceling each other out. At $t \approx 7$ ms the individual contributions of the PBFs stop canceling themselves out: they add up to create the re-entering characteristic wave. This stresses how the modeling procedure of $R(\omega)$ is critical for an accurate prediction of acoustic delays.

5. Validation for a combustion chamber

In this section, we investigate the thermoacoustic instability [45] in a laminar experimental setup called INTRIG Burner (IMFT, Toulouse). First, a DNS of the full setup (called “FULL”) is conducted. A second DNS (called “TRUNCATED”) in which the computational domain is truncated after the flame is conducted. In the TRUNCATED case, D-TDIBC is used to model the acoustic properties of the truncated domain as in Sec. 4. The results obtained using D-TDIBC will be compared with those of the FULL configuration.

5.1. Experimental setup

The INTRIG Burner is used to study a lean premixed laminar methane–air flame attached on a cylinder. The operating point corresponds to an equivalence ratio $\phi = 0.75$ and a bulk velocity of the fresh gases of $u_b = 0.8 \text{ m s}^{-1}$. The associated laminar flame speed and adiabatic temperature are $s_l = 0.23 \text{ m s}^{-1}$ and $T_{ad} = 1920 \text{ K}$.

The experimental rig is shown in Fig. 10. The gaseous methane–air premixture is injected upstream of the glass ball array located at $x = -0.367 \text{ m}$. The flow is then laminarized by the glass balls and the honey comb panels. A lean premixed methane–air laminar flame then attaches to a cylindrical stainless-steel flame holder of with a diameter $d = 8 \text{ mm}$. The combustion chamber has a constant cross section of $h = 34 \text{ mm}$ by $l = 94 \text{ mm}$.

5.2. Numerical setup

As the flow is two-dimensional, 2D Direct Numerical Simulation (DNS) approach can be used [46–48]. A 19-species mechanism (called LU19) is used [49] for the modeling of chemical kinetics. Schmidt and Prandtl numbers are assumed constant: $\text{Pr} = 0.6$ and $\text{Sc} = 0.6$. The full mesh is composed of 625 000 cells. The reduced mesh is composed of 545 000 cells, i.e. a reduction of 13%. The flame thickness is $\delta_L^0 = 680 \text{ }\mu\text{m}$ and at least 11 points are used to resolve the flame front with a mesh resolution at the flame of $\Delta x = 60 \text{ }\mu\text{m}$ [46–48]. A typical flame, here at a stable operating point, is shown in Fig. 10 (bottom).

The inlet acoustic boundary (glass balls at $x = -0.367 \text{ m}$) is a hard wall (i.e. $u(\omega) = 0 \text{ m s}^{-1}$ at all frequencies). It was checked in the experiment that this is a good approximation [48]. This assumption is close to realistic conditions as the inlet BC corresponds to the glass balls array. The outlet acoustic boundary (at $x = 0.35 \text{ m}$) is an open-end (i.e. $p(\omega) = 0 \text{ Pa}$ at all frequencies). This issue should be taken care of if one were willing to compute accurately the thermoacoustic stability of the

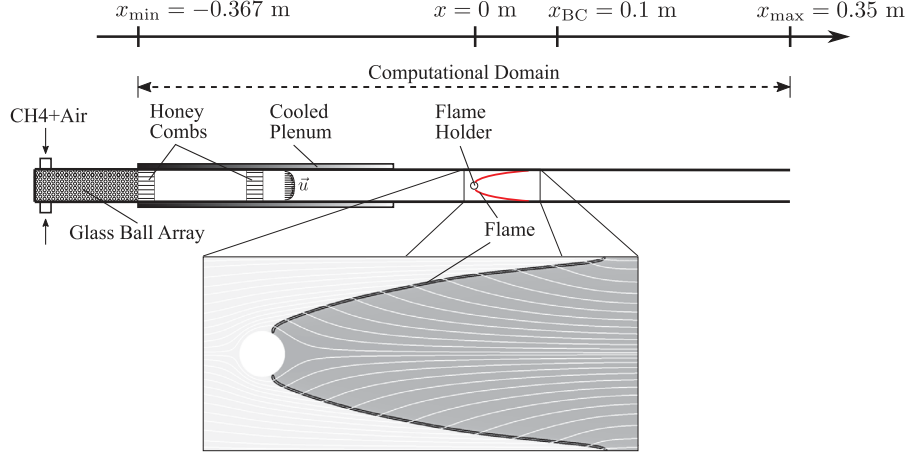


Fig. 10. Sketch of a transverse cut of the INTRIG Burner (top). The computational domain of the FULL configuration goes from the glass balls array ($x = -0.367$ m) to the exhaust ($x = 0.35$ m) while it is cut (at $x = x_{BC} = 0.1$ m) in the TRUNCATED configuration. A zoom (bottom) of the flame region of the initial solution used in DNS is shown. The light grey and dark grey regions represent the zero and high CO_2 mass fraction levels. The streamlines are plotted in white.

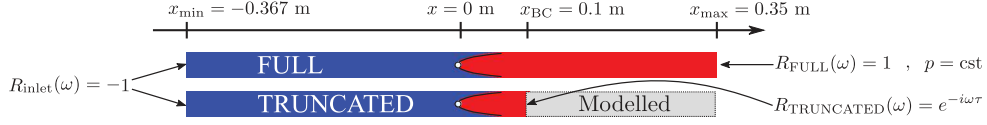


Fig. 11. Schematic of the computational domains and BC used for the two simulations: FULL and TRUNCATED cases.

Table 1

Parameters used in the two simulations.

Case	Domain length	N_{cell}	Δx_{\min}	Inlet BC	Outlet BC	Outlet BC method
FULL	0.717 m	625000	60 μm	$R(\omega) = -1$	$R_{\text{FULL}}(\omega) = 1$	NSCBC
TRUNCATED	0.467 m	545000	60 μm	$R(\omega) = -1$	$R_{\text{TRUNCATED}}(\omega) = e^{-i\omega\tau}$	D-TDIBC

configuration. However, the goal of this paper is more modest: it aims to validate D-TDIBC. The simple acoustic conditions used here are thus acceptable as the reference case is a numerical simulation.

Two simulations are carried out and investigated in this section (Fig. 11): the “FULL” and the “TRUNCATED” cases. The outlet boundary condition of the FULL domain is located as $x = 0.35$ m and the outlet D-TDIBC boundary condition of the TRUNCATED domain is located at $x = 0.1$ m. Fig. 11 shows the computational domain and the boundary conditions used in the three simulations. The grey area in the TRUNCATED case corresponds to the domain where the acoustic wave propagation is modeled by a time delay imposed by D-TDIBC. Table 1 summarizes the parameters used in the simulation.

The truncation of the gray domain in Fig. 11 is possible as the flow is strongly one-dimensional and homogeneous. For instance, the axial velocity is 4 orders of magnitude higher than the maximum of the vertical velocity in the truncated portion.

5.3. D-TDIBC: reflection coefficient model

The Iterative Multi-Pole Modeling Technique presented in Sec. 3 is used to model the acoustic wave propagation in the truncated part of the INTRIG Burner. The outlet boundary conditions of the Full and Truncated domains are located at $x_{\max} = 0.35$ m and $x_{BC} = 0.10$ m, respectively. Consequently, the length of the domain to be modeled by D-TDIBC is $L = 0.25$ m. The value of x_{BC} must be large enough to ensure that combustion is complete and only acoustics take place between x_{BC} and x_{\max} . Moreover, the sound speed must be homogeneous in this zone. This condition has been verified *a posteriori* by analyzing the results of the FULL configuration simulation: the sound speed fluctuations are less than 0.5% of the mean value $c_0 = 830$ m/s. According to duct acoustics theory, plane waves propagate in ducts at frequencies lower than a cutoff frequency f_c [34]. This frequency is given by the speed of sound c_0 and the lowest height in the cross section. In the INTRIG setup, the plane waves propagate at frequencies lower than $f_c = 3.5$ kHz. The filtering procedure used in (Sec. 4.2) is applied here so that the frequencies lower than $f_c = 3.5$ kHz are accurately modeled. The results are shown in Fig. 12. The reflection coefficient model used here consists in a set of 18 (p_k, μ_k) and verifies causality, reality and passivity conditions (cf. Appendix B).

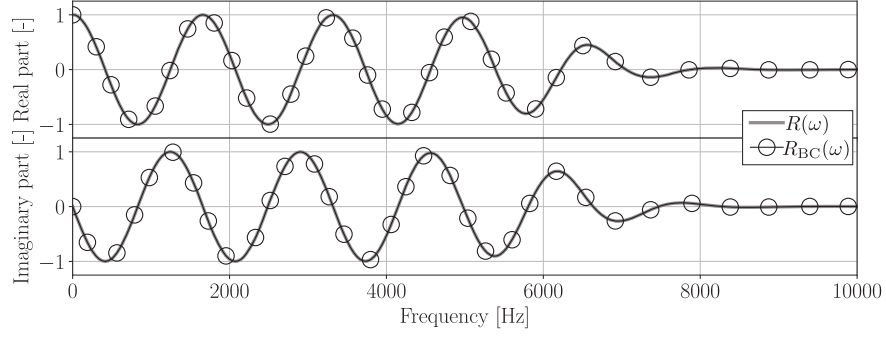


Fig. 12. Model of the delayed reflection coefficient $R_{BC}(\omega)$ with $n_0 = 18$ (see Eq. (13)) for a time-delay of $\tau = 0.30$ ms in the laminar combustor of Fig. 10.

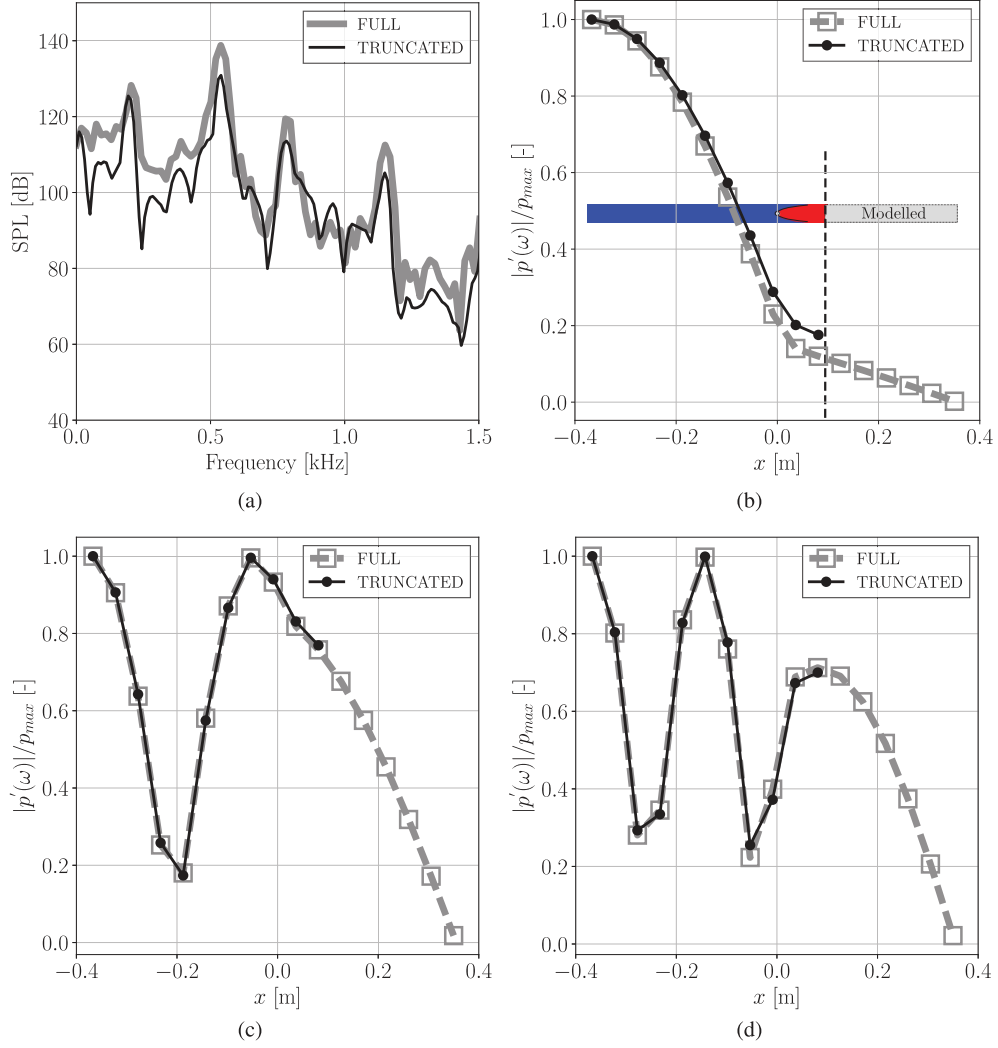


Fig. 13. (a) Spectra obtained for the FULL and the TRUNCATED simulations. D-TDIBC allows an accurate prediction of the thermoacoustic stability observed in the FULL configuration. The spectra are based on a probe located at $x = -8.5$ mm. The shapes of the first (b), second (c) and third (d) modes correspond to the acoustic eigenmodes of a left-hand side closed and right-hand side open duct modified by the jump of characteristic impedance between the fresh and burnt gases. The markers correspond to the probes in the simulation (lines are added for the sake of clarity).

5.4. Results and discussion

The flow setup in Fig. 10 is thermoacoustically unstable: the flame oscillates and couples to the acoustic modes of the setup [45]. The instability is sustained in DNS of the FULL and TRUNCATED configurations (Fig. 11). Fig. 13(a) illustrates the

Sound Pressure Level (SPL) spectra expressed in dB and defined as $SPL = 20 \log(p/p_{\text{ref}})$ based on a reference pressure level of $p_{\text{ref}} = 2 \cdot 10^{-5}$ Pa. The FULL simulation, i.e. computing the full domain, shows a strong acoustic activity with many amplified acoustic modes. Four principle modes are found, with frequencies 203 Hz, 534 Hz, 782 Hz and 1150 Hz, respectively. The mode shapes can be obtained by (1) performing a spectral analysis at several axial locations of the burner, (2) extracting the amplitude of the Fourier transform at the peak frequency in Fig. 13(a) and (3) normalizing by the maximum pressure amplitude. The three first modes observed in the FULL simulation are a quarter wave mode (Fig. 13(b)), a three-quarter wave mode (Fig. 13(c)) and a five-quarter wave mode (Fig. 13(d)).

The TRUNCATED simulation also exhibits thermoacoustic instabilities and is in good agreement with the FULL simulation. Fig. 13(a) illustrates the ability of D-TDIBC to accurately predict the broadband acoustic activity, despite the clear nonlinear excitation of higher harmonics, due to the high pressure amplitudes levels. In particular, the agreement between values of the excited frequencies (Fig. 13(a)) and the mode shapes (Figs. 13(b), (c) and (d)) is remarkable.

Some discrepancies are visible in the magnitude of the SPL spectrum, where the TRUNCATED simulations overall under-predict the intensity of the SPLs. This discrepancy can be attributed to two concurrent issues. First, the reflection coefficient used at the outlet of the FULL simulation is $|R_{\text{FULL}}| = 1$ for all frequencies, while this is true only in a selected frequency range of interest (due to the filtering procedure used, cf. Fig. 12 and Fig. 6(b)) for the TRUNCATED simulation. Outside of that range, the magnitude of the reflection coefficient drops below 1, hence causing spurious acoustic energy loss at the D-TDIBC boundary. Second, the large amplitude of the pressure oscillations at the limit cycle may trigger losses due to non-linear acoustic effects which are directly simulated in the FULL simulations in a portion of the domain that is not present in the TRUNCATED simulations.

Overall, Fig. 13 demonstrates that D-TDIBC can satisfactorily recover the acoustic eigenmodes of a domain while simulating only a part of it. It also gives the opportunity to investigate the thermoacoustic stability of different setups while simulating the same computational domain: the length of the truncated domain can be adjusted within the same DNS simply by introducing a different time delay at the boundary $x = x_{\text{BC}}$ (Fig. 10), without the need to create a new computational mesh.

6. Conclusion

A novel numerical strategy called Delayed-Time Domain Impedance Boundary Condition (D-TDIBC) has been derived from an existing TDIBC approach [1–4] in order to account for time delays in Navier–Stokes simulation. The method allows to mimic acoustic wave propagation in domains larger than the computational domain used in the numerical study. The benefits are twofold: (1) computational cost can be saved and spatial resolution can focus on areas of interest, (2) any additional length can be added to the computational domain allowing straightforward investigations of geometry changes downstream or upstream of the zone of interest. While the former mainly concerns structured meshes, the latter applies to all types of grids.

D-TDIBC is an extension of Time Domain Impedance Boundary Condition (TDIBC) methods [1–4]. The proposed methodology works with the reflection coefficient $R(\omega)$ and gives accurate models for time delayed fully reflecting acoustic boundary conditions such as closed and open-end conditions. These models were used in the time domain simulations and were found to give accurate acoustic wave propagation properties for both one-dimensional and two-dimensional reactive Navier–Stokes simulations.

The one-dimensional wave propagation results illustrate the capacity of D-TDIBC to accurately impose a time delay while conserving the acoustic energy in case of a full reflection. The investigation on individual Pole Base Functions (PBFs) stresses the need for an accurate modeling strategy. Indeed, the time delayed response of the boundary results in a cancellation of individual responses: the PBFs. This is accomplished through the diversity of their amplitudes, pseudo-periods and phases.

The two-dimensional simulation focuses on the thermoacoustic stability of a laminar combustion chamber (the INTRIG Burner [46–48]) with a lean premixed laminar methane–air flame attached on a cylinder. A portion of the outlet domain was truncated and the acoustic wave propagation was modeled with D-TDIBC. The D-TDIBC simulation was compared to a reference simulation where the full domain is taken into account. D-TDIBC was found to give accurate predictions of the thermoacoustic stability while saving computational resources. The frequencies, amplitudes and mode shapes are in excellent agreement with the reference simulation.

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Appendix A. D-TDIBC model: 1D wave propagation problem (Table A.2)

Table A.2

Poles and residues used in the one-dimensional pure delayed Gaussian wave propagation problem.

R_k	a_k	b_k	c_k	d_k	$\omega_{0,k}$ [rad s ⁻¹]	f_0 [Hz]
$k = 1$	4.91e+05	4.56e+05	-3.23e+03	3.48e+03	4.75e+03	7.56e+02
$k = 2$	-1.10e+04	-7.38e+03	-9.52e+02	1.42e+03	1.71e+03	2.72e+02
$k = 3$	-8.21e+04	-3.92e+04	-1.38e+03	2.89e+03	3.20e+03	5.09e+02
$k = 4$	-1.33e+05	-4.66e+04	-1.50e+03	4.29e+03	4.54e+03	7.23e+02
$k = 5$	-6.17e+04	-1.39e+04	-1.30e+03	5.79e+03	5.93e+03	9.44e+02
$k = 6$	-3.22e+04	-5.12e+03	-1.15e+03	7.26e+03	7.35e+03	1.17e+03
$k = 7$	-1.73e+04	-2.04e+03	-1.03e+03	8.73e+03	8.79e+03	1.40e+03
$k = 8$	-1.19e+02	2.65e+07	-2.17e+02	-9.77e-04	2.17e+02	3.46e+01
$k = 9$	-2.11e+00	6.77e+04	-1.89e+00	-5.88e-05	1.89e+00	3.01e-01
$k = 10$	-9.15e+03	-8.40e+02	-9.34e+02	1.02e+04	1.02e+04	1.63e+03
$k = 11$	-1.32e+05	-1.15e+05	-7.95e+03	9.11e+03	1.21e+04	1.92e+03
$k = 12$	6.55e+02	5.49e+02	-5.20e+02	6.20e+02	8.09e+02	1.29e+02
$k = 13$	-5.04e+03	-4.09e+02	-9.40e+02	1.16e+04	1.16e+04	1.85e+03
$k = 14$	-2.25e+03	-1.70e+02	-9.78e+02	1.30e+04	1.30e+04	2.07e+03
$k = 15$	-3.80e+01	-8.77e+07	-7.84e+01	3.40e-05	7.84e+01	1.25e+01
$k = 16$	-7.67e+03	-9.34e+02	-1.80e+03	1.48e+04	1.49e+04	2.37e+03
$k = 17$	6.89e+00	1.17e-01	-1.61e+02	9.46e+03	9.46e+03	1.51e+03
$k = 18$	-1.50e+01	1.75e+06	-2.93e+01	-2.51e-04	2.93e+01	4.67e+00
$k = 19$	3.82e+03	3.80e+02	-1.50e+03	1.50e+04	1.51e+04	2.40e+03
$k = 20$	-6.06e+00	-7.54e+05	-9.40e+00	7.56e-05	9.40e+00	1.50e+00

Appendix B. D-TDIBC model: 2D reactive DNS of the INTRIG Burner (Table B.3)

Table B.3

Poles and residues used in the two-dimensional reactive DNS of the INTRIG Burner.

R_k	a_k	b_k	c_k	d_k	$\omega_{0,k}$ [rad s ⁻¹]	f_0 [Hz]
$k = 1$	1.91e+05	-5.06e+04	-8.06e+03	-3.04e+04	3.14e+04	5.00e+03
$k = 2$	-3.24e+06	-1.10e+07	-1.83e+04	5.39e+03	1.91e+04	3.04e+03
$k = 3$	1.47e+06	1.97e+06	-1.17e+04	8.74e+03	1.46e+04	2.32e+03
$k = 4$	5.20e+05	2.55e+05	-9.61e+03	1.96e+04	2.18e+04	3.47e+03
$k = 5$	1.33e+03	-3.60e+09	-2.74e+03	-1.01e-03	2.74e+03	4.37e+02
$k = 6$	5.06e+02	-1.41e+09	-1.06e+03	-3.81e-04	1.06e+03	1.69e+02
$k = 7$	7.45e+05	1.08e+06	-3.97e+04	2.75e+04	4.83e+04	7.69e+03
$k = 8$	-2.45e+03	-2.62e+02	-3.78e+03	3.54e+04	3.56e+04	5.66e+03
$k = 9$	3.15e+04	4.45e+03	-5.68e+03	4.02e+04	4.06e+04	6.46e+03
$k = 10$	6.28e-03	-1.01e+05	-6.28e-03	-3.93e-10	6.28e-03	1.00e-03
$k = 11$	2.16e+02	-3.31e+07	-3.85e+02	-2.51e-03	3.85e+02	6.13e+01
$k = 12$	1.86e+01	2.39e-03	-1.08e-02	8.46e+01	8.46e+01	1.35e+01
$k = 13$	7.04e-01	3.79e-14	-4.51e-12	8.38e+01	8.38e+01	1.33e+01
$k = 14$	5.11e+03	6.29e+02	-6.01e+03	4.88e+04	4.92e+04	7.83e+03
$k = 15$	2.66e+05	1.33e+05	-2.10e+04	4.20e+04	4.70e+04	7.47e+03
$k = 16$	1.89e+01	2.85e+05	-6.39e+01	4.24e-03	6.39e+01	1.02e+01
$k = 17$	6.76e+03	1.68e+03	-5.56e+03	2.23e+04	2.30e+04	3.66e+03
$k = 18$	5.92e+01	-6.50e+01	-1.14e+02	-1.04e+02	1.54e+02	2.45e+01

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